

C-130 Usage/Environmental Criteria Analysis and Gust Loads Assessment

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Outline

- C-130 SLAP Background, Requirements, and Objectives
- Gust Clustering Effects (Mr. McColl)
- Environmental Criteria Development (Dr. Moon)
 - Maneuver
 - Gust
 - Taxi
 - Landing Impact
- Conclusions

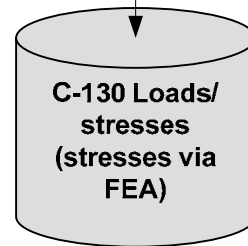
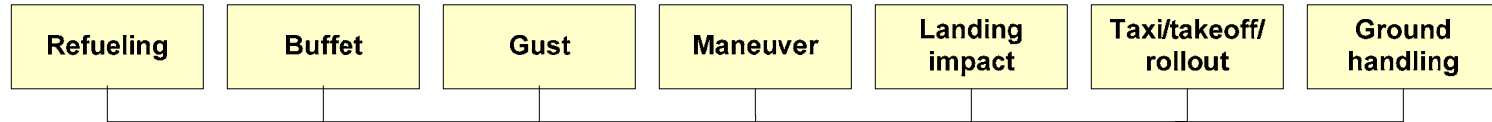
US Navy C-130 Service Life Assessment (SLAP)

Why perform a SLAP for the C-130?

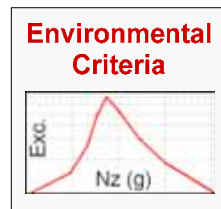
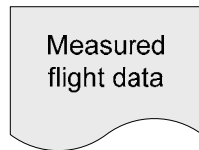
- Inspections reveal fatigue cracking at various locations
- Grounded (or potentially grounded) aircraft
- C-130 usage continues to change and evolve
- Effects due to individual aircraft usage
- Limited aircraft resources and increased demand requires optimized fleet management and aircraft life assessments
- These requirements dictate the need for the following:
 - More refined representations of fleet usage/environmental criteria
 - Analyze (K)C-130F/R/T/J recorded data to create Nz maneuver, Nz gust, VGH-VH, mission mix, mission profile, maneuver sequence (with time, weight, fuel, cargo, velocity, and altitude), and landing impact/ground maneuver criteria
 - Structural integrity assessments
- The focus of this presentation is environmental criteria – specifically the most critical load sources (gust and maneuver); ground and landing impact are addressed as well

SLAP: Areas of Interest

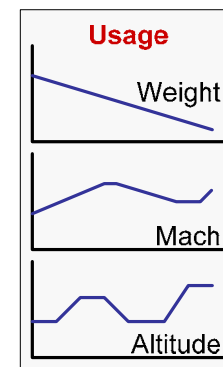
Load sources



Focus of this presentation

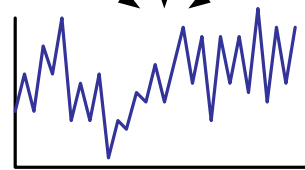


Mission profiles
Mission mix



- SDRS - (K)C-130 F/R/T**
 - 132,000+ hours data
 - Event-based recording
 - Captures maneuver, storm gust
- GMS - KC-130J**
 - 18,000+ hours data
 - High sample rate
 - Continuous recording
 - Captures all load sources

Predicted Stress History



Fatigue/DTA

**Damage, crack growth,
risk assessment**

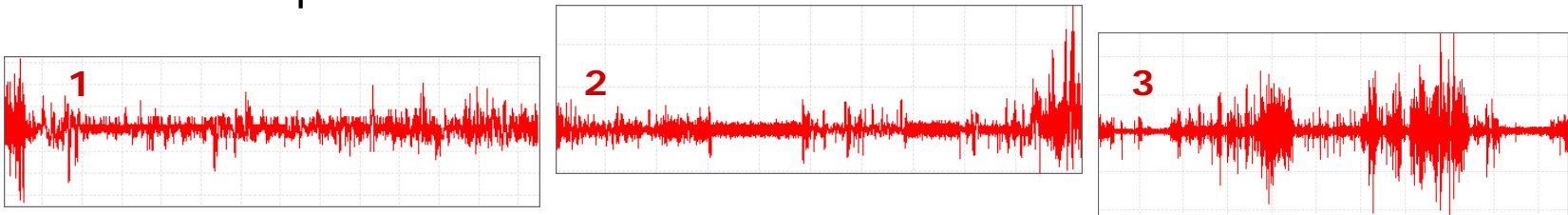
Gust Cycle Clustering Effects



Photo source: "Hurr Katrina Aug 05-005.jpg" (<http://www.aoc.noaa.gov>)

Gust Cycle Clustering Effects

- Aircraft fatigue and damage tolerance analyses are highly sensitive to the placement of large amplitude cycles within the underlying stress sequence (spectra)
- Large amplitude cycles occurring near the beginning¹ of the stress sequence can potentially result in large residual stresses and, subsequently, slower crack growth or even crack retardation
- Conversely, large amplitude cycles occurring near the end² of the stress sequence can result in faster crack growth due to the absence of large residual stresses earlier in the spectrum
- A variety of intermediate effects can be achieved by more uniform distributions³ of large amplitude cycles throughout the stress sequence



Gust Cycle Clustering Effects

- For aircraft instrumented with high sample rate data recorders where fatigue is tracked on a flight-by-flight basis, these large amplitude cycles can be measured and applied in analysis exactly where/when they occur
 - KC-130J GMS recorder
- For other aircraft, such information may not be directly known
 - Also, for more generalized fleet-wide studies (i.e., mission analysis or airframe fatigue test spectra generation), such information may also not be directly known
- Where should cycles be placed?
 - Pilot-induced maneuvers: typically distributed evenly across such a spectrum (based on mission type/flight segment)
 - Gust-induced cycles (e.g., MIL-based power spectral density (PSD) gust): typically formulated based purely on altitude and time-in-segment, with no known information provided as to where to explicitly place cycles across the sequence

Gust Cycle Clustering Effects

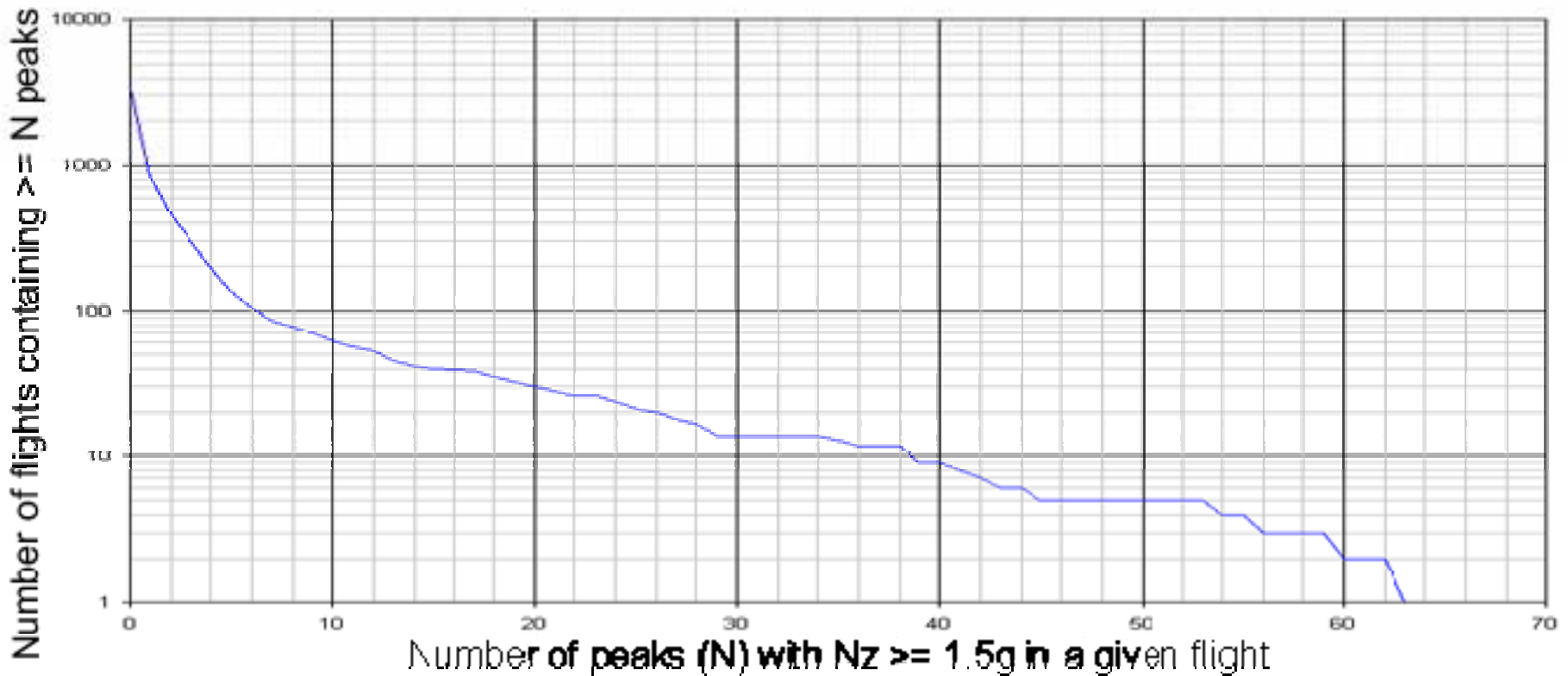
- Typical gust cycle placement
 - Evenly distribute large amplitude gust cycles across the sequence
 - Inconsistent with the randomness of turbulence (a random process should display no pattern or regularity)
 - Group them in a subset of flights in some random fashion
 - Wide variation in crack growth rates can result based on where the high amplitude cycles were randomly placed
- KC-130J measured flight data was examined to better model the placement of gust cycles across a sequence
- Two objectives
 - Examine the distribution of the number of gust cycles per flight
 - Examine the gust cycle clustering effects, wherein high amplitude gust cycles tend to congregate together in a subset of flights across the spectrum
 - Significant in that it provides insight into how to better model the random nature of turbulence when developing spectra for fatigue and damage tolerance analysis

Gust Cycle Clustering Effects

- Previous work by Bullen* states
 - "The application of a given number of loading cycles between each ground-to-air cycle is unrealistic. Some flights are calm, some are extremely turbulent, and the majority of flights range somewhere between the two extremes."*
- The objective is to quantify this statement
- First to be determined was the distribution of the number of peaks at-or-above a given g-level across flights
 - i.e., how uniform was the distribution of gust cycles flight-to-flight

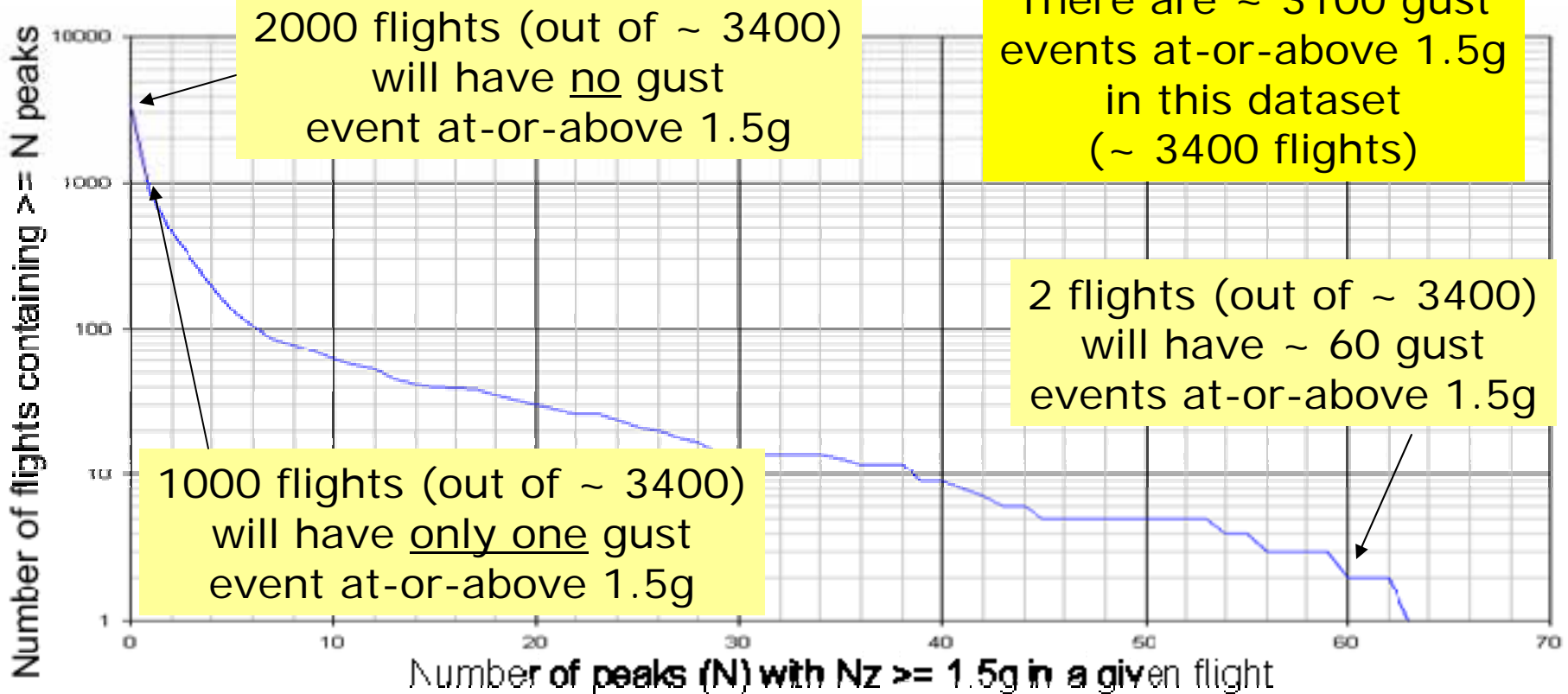
Gust Cycle Clustering Effects

- Results show a broad distribution of the number of gust cycles across flights (1.5g example)



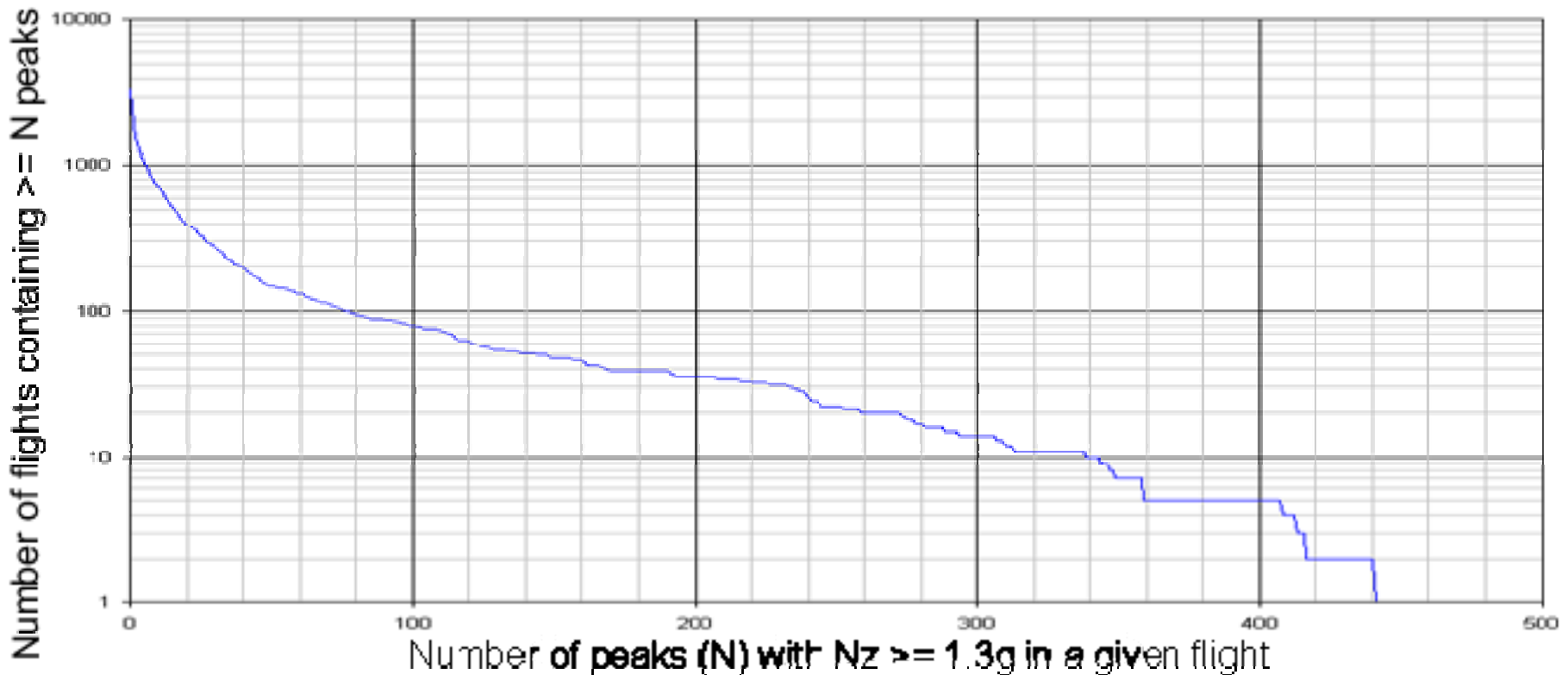
Gust Cycle Clustering Effects

- Results show a broad distribution of the number of gust cycles across flights (1.5g example)



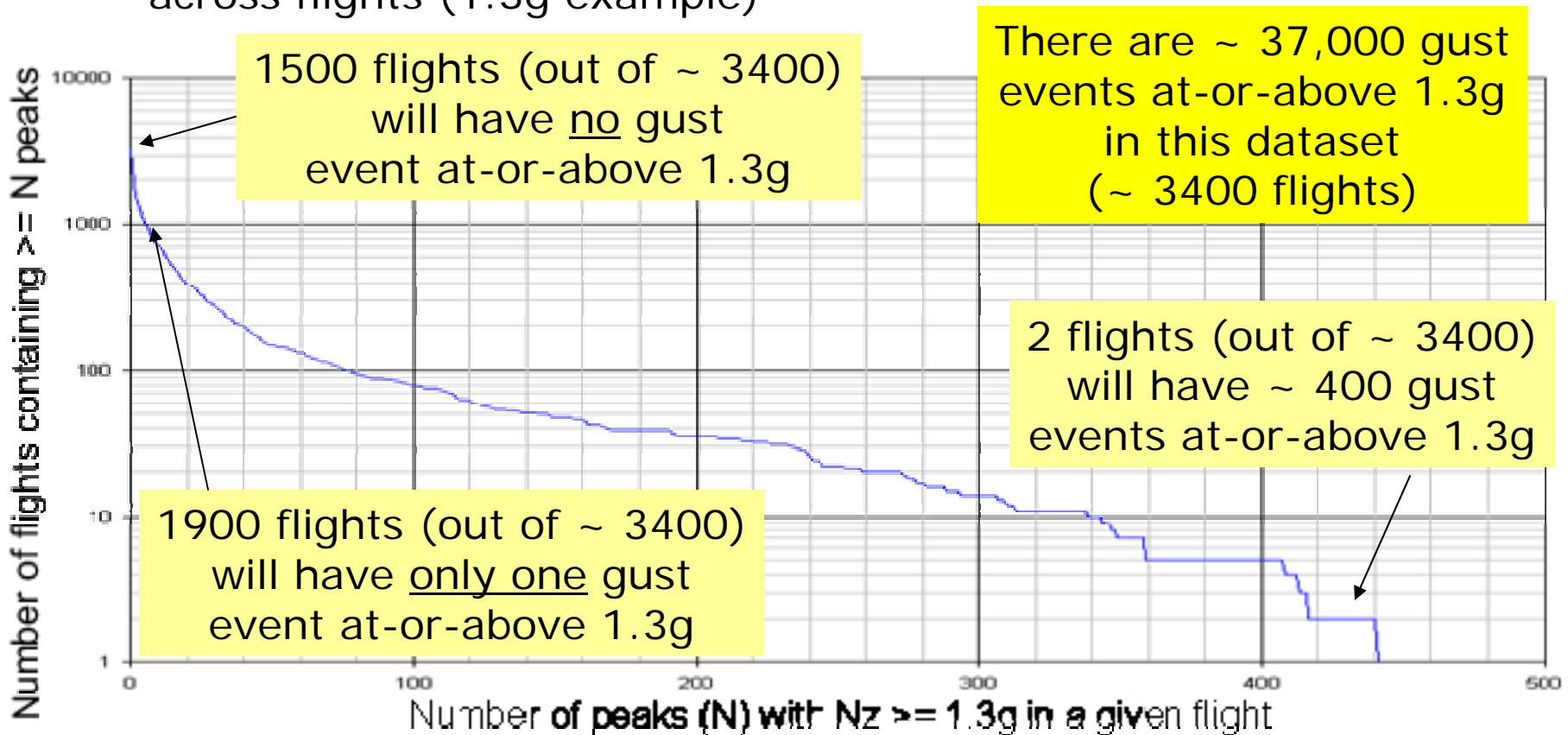
Gust Cycle Clustering Effects

- Results show a broad distribution of the number of gust cycles across flights (1.3g example)



Gust Cycle Clustering Effects

- Results show a broad distribution of the number of gust cycles across flights (1.3g example)



Gust Cycle Clustering Effects

- For higher g-levels (with more isolated events), the distribution tends to converge
 - 1.7g:
 - Occurs in ~7% of flights
 - Multiple cycles per flight occur across 30% these flights
- This is based in-part on limited dataset, yet likely still a real phenomenon
- However, fatigue and crack growth can be extremely sensitive to these abundantly occurring “mid” g-level gust events, so it is critical to spread them appropriately

Gust Cycle Clustering Effects

- Next to be determined were gust cycle clustering effects, wherein high amplitude gust cycles potentially congregate together in a subset of flights across the spectrum
- Gust occurrences due to discrete g-levels (1.2, 1.3, ..., 1.9g) were determined
 - *Gust cycles $\geq 2.0g$ were examined as well, but determined to be too small a dataset*
- Correlation (measure of linearity between two quantities) and significance were calculated across g-levels within flights
- Also determined was the presence of lower-g events clustered together within each flight
- The general trend (as seen in following two charts) is that the flights with the higher g-counts tend to have more overall gust cycles within that flight; i.e., gust cycles tend to “cluster” together

Air Vehicle Engineering

Correlation coefficient (ρ_{ij}) matrix

	nz_1.1	nz_1.2	nz_1.3	nz_1.4	nz_1.5	nz_1.6	nz_1.7	nz_1.8	nz_1.9	nz_2	nz_2.1	nz_2.2	nz_2.3	nz_2.4
nz_1.1		0.6749	0.575	0.506	0.4652	0.3815	0.3137	0.2067	0.1838	0.092	0.0601	0.0587	0.1033	-0.0033
nz_1.2	0.6749		0.9538	0.8834	0.8033	0.6879	0.5376	0.3965	0.2853	0.1672	0.1011	0.0884	0.072	-0.0047
nz_1.3	0.575	0.9538		0.9603	0.8917	0.7765	0.606	0.4691	0.315	0.1796	0.1057	0.0985	0.0964	-0.0047
nz_1.4	0.506	0.8834	0.9603		0.9251	0.8199	0.654	0.516	0.3126	0.2037	0.088	0.121	0.1062	-0.0025
nz_1.5	0.4652	0.8033	0.8917	0.9251		0.8182	0.6674	0.574	0.3397	0.2146	0.0953	0.1496	0.1035	-0.0042
nz_1.6	0.3815	0.6879	0.7765	0.8199	0.8182		0.5908	0.6028	0.344	0.2295	0.0681	0.1236	0.0632	-0.0036
nz_1.7	0.3137	0.5376	0.606	0.654	0.6674	0.5908		0.5246	0.2498	0.1668	0.0683	0.1241	0.0808	0.0404
nz_1.8	0.2067	0.3965	0.4691	0.516	0.574	0.6028	0.5246		0.3269	0.2074	0.0514	0.1573	0.0363	-0.002
nz_1.9	0.1838	0.2853	0.315	0.3126	0.3397	0.344	0.2498	0.3269		0.033	0.0527	0.1579	0.1579	-0.0015
nz_2	0.092	0.1672	0.1796	0.2037	0.2146	0.2295	0.2498	0.269	0.2974	0.3397	0.3965	0.4691	0.516	-0.0024
nz_2.1	0.0601	0.1011	0.1057	0.1062	0.1095	0.1236	0.1241	0.1269	0.133	0.1668	0.2074	0.2498	0.2974	0.1754
nz_2.2	0.0587	0.0884	0.0985	0.121	0.1496	0.1668	0.1838	0.2067	0.2295	0.269	0.3269	0.3965	0.4691	-0.0012
nz_2.3	0.1033	0.072	0.0964	0.1062	0.1035	0.0632	0.0808	0.0363	0.1579	0.1579	0.1579	0.1579	0.1579	-0.0006
nz_2.4	-0.0033	-0.0047	-0.0047	-0.0025	-0.0042	-0.0036	0.0404	-0.002	-0.0015	-0.0012	-0.0008	-0.0006	-0.0006	-0.0006

Too small a dataset

Good correlation; good significance

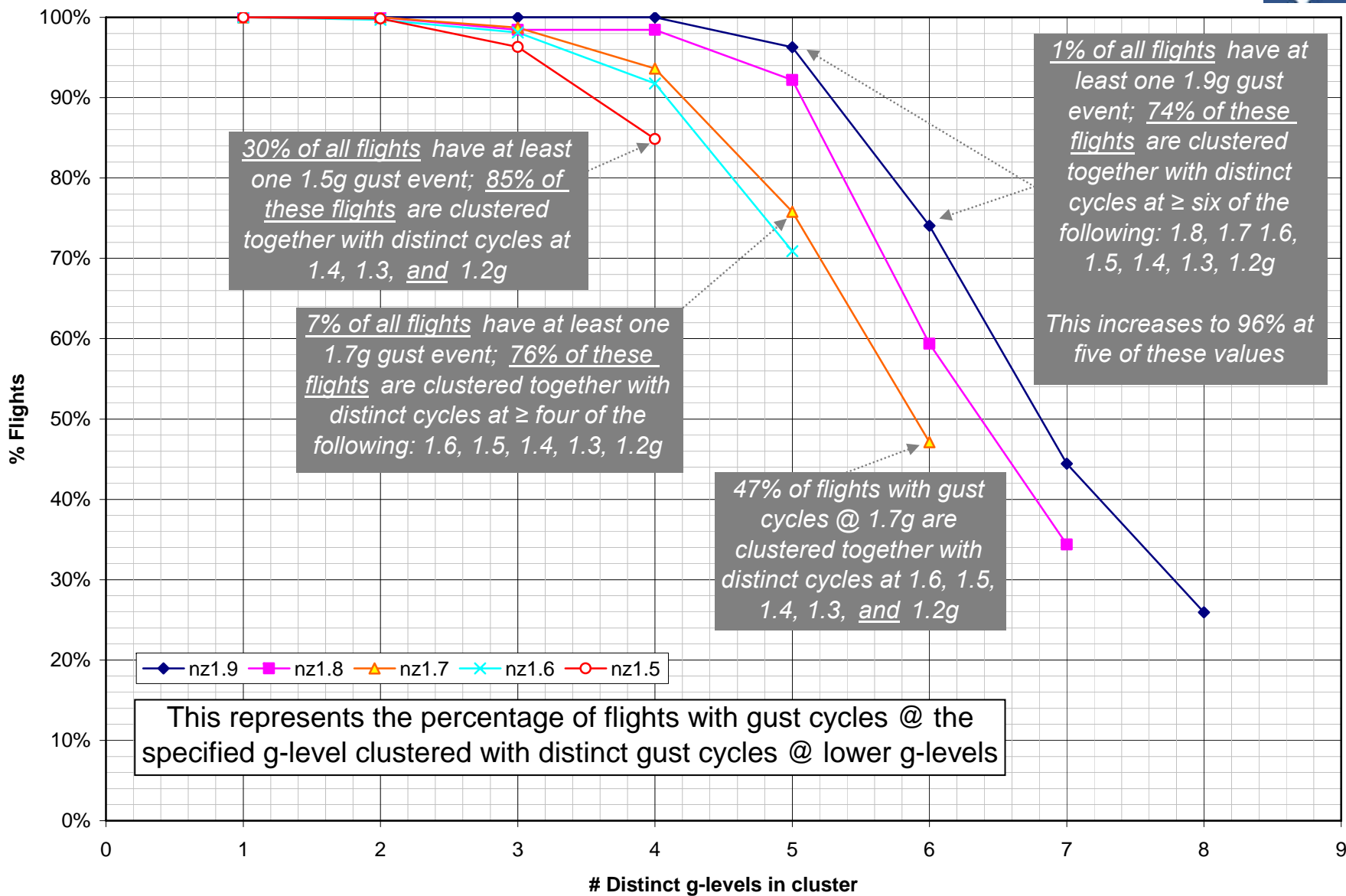
p-value* matrix

	nz_1.1	nz_1.2	nz_1.3	nz_1.4	nz_1.5	nz_1.6	nz_1.7	nz_1.8	nz_1.9	nz_2	nz_2.1	nz_2.2	nz_2.3	nz_2.4
nz_1.1		0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.1%	0.0%	84.6%
nz_1.2	0.0%		0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	78.3%
nz_1.3	0.0%	0.0%		0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	78.3%
nz_1.4	0.0%	0.0%	0.0%		0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	88.4%
nz_1.5	0.0%	0.0%	0.0%	0.0%		0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	80.7%
nz_1.6	0.0%	0.0%	0.0%	0.0%	0.0%		0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	83.3%
nz_1.7	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%		0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	1.8%
nz_1.8	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%		0.0%	0.0%	0.3%	0.0%	3.4%	90.8%
nz_1.9	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%		5.4%	0.2%	0.0%	0.0%	93.2%
nz_2	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	5.4%		84.1%	88.7%	88.7%	94.4%
nz_2.1	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	84.1%		88.7%	88.7%	96.1%
nz_2.2	0.1%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	88.7%		88.7%	97.3%
nz_2.3	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	0.0%	88.7%		97.3%
nz_2.4	84.6%	78.3%	78.3%	88.4%	80.7%	83.3%	1.8%	90.8%	93.2%	94.4%	96.1%	97.3%	97.3%	

Too small a dataset

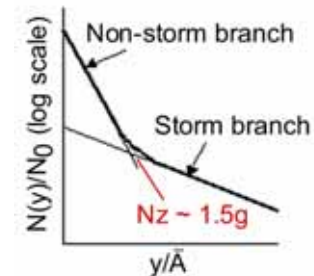
* The **p-value** (p) is the probability of getting a correlation as large as the observed value by random chance, when the true correlation is zero; for small p (less than a 5% significance level is standard) the correlation ρ_{ij} is significant

Air Vehicle Engineering

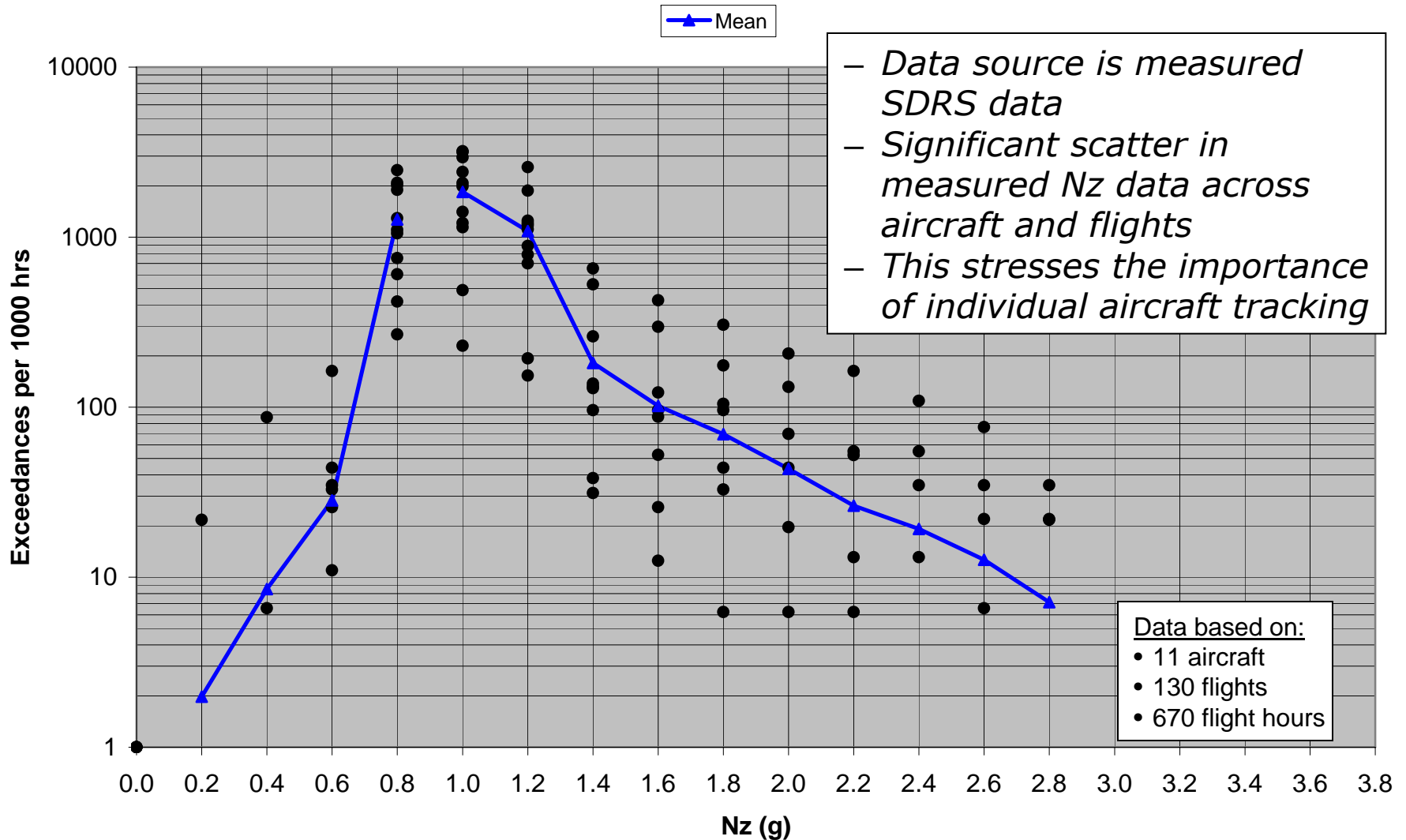


C-130 Environmental Criteria

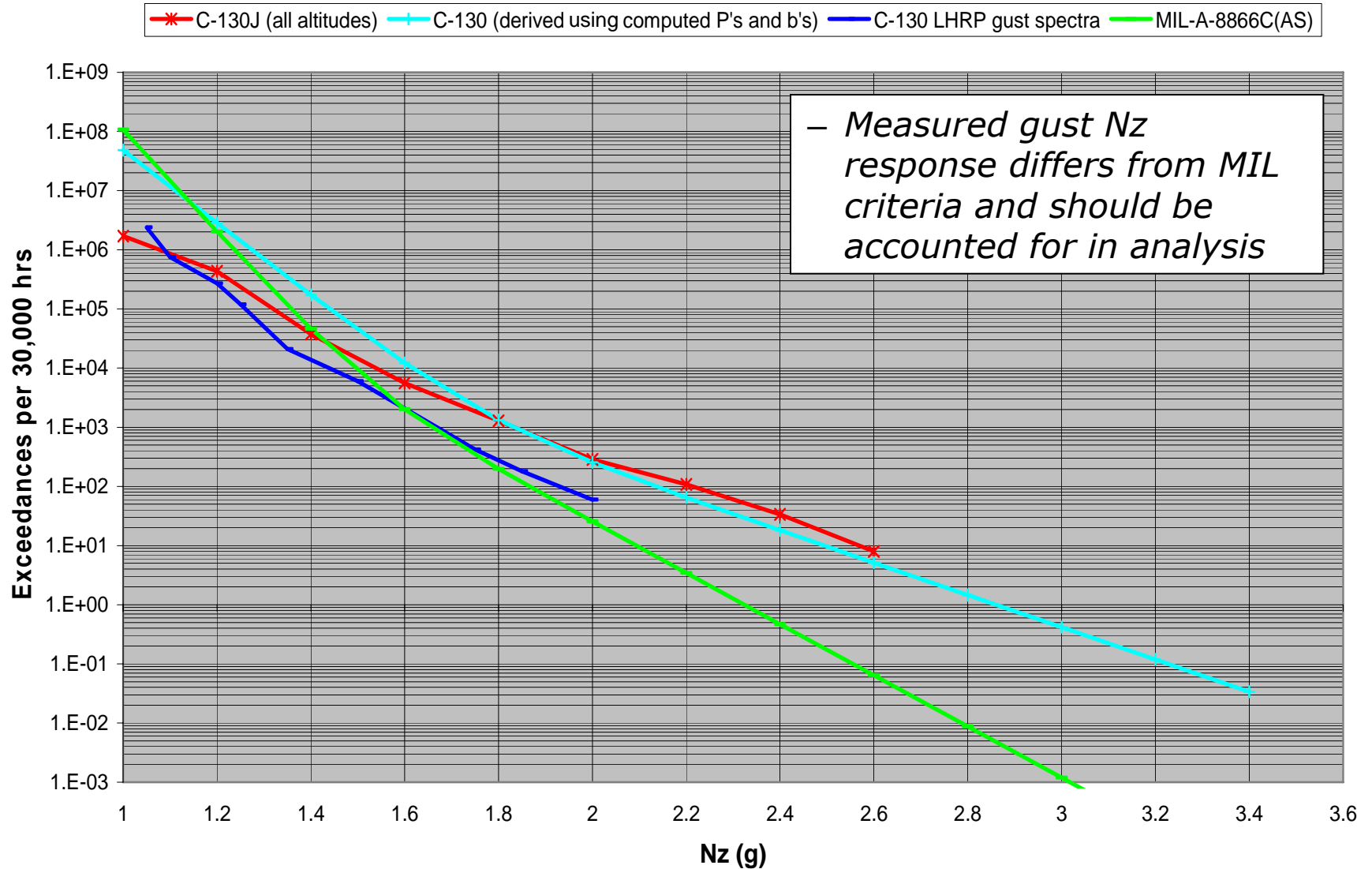
- Maneuver and gust criteria for the KC-130 F/R/T and C-130T is based on measured SDRS data
 - 132,000+ hours data
 - Event-based recording
 - Captures maneuver, storm gust
- Maneuver, gust, and ground criteria for the KC-130J is based on measured GMS data
 - 18,000+ hours data
 - High sample rate
 - Continuous recording
 - Captures all load sources
- Landing impact criteria is based on legacy analysis compared with recent measured test data



KC-130R Transport Mission Maneuver Nz Exceedances

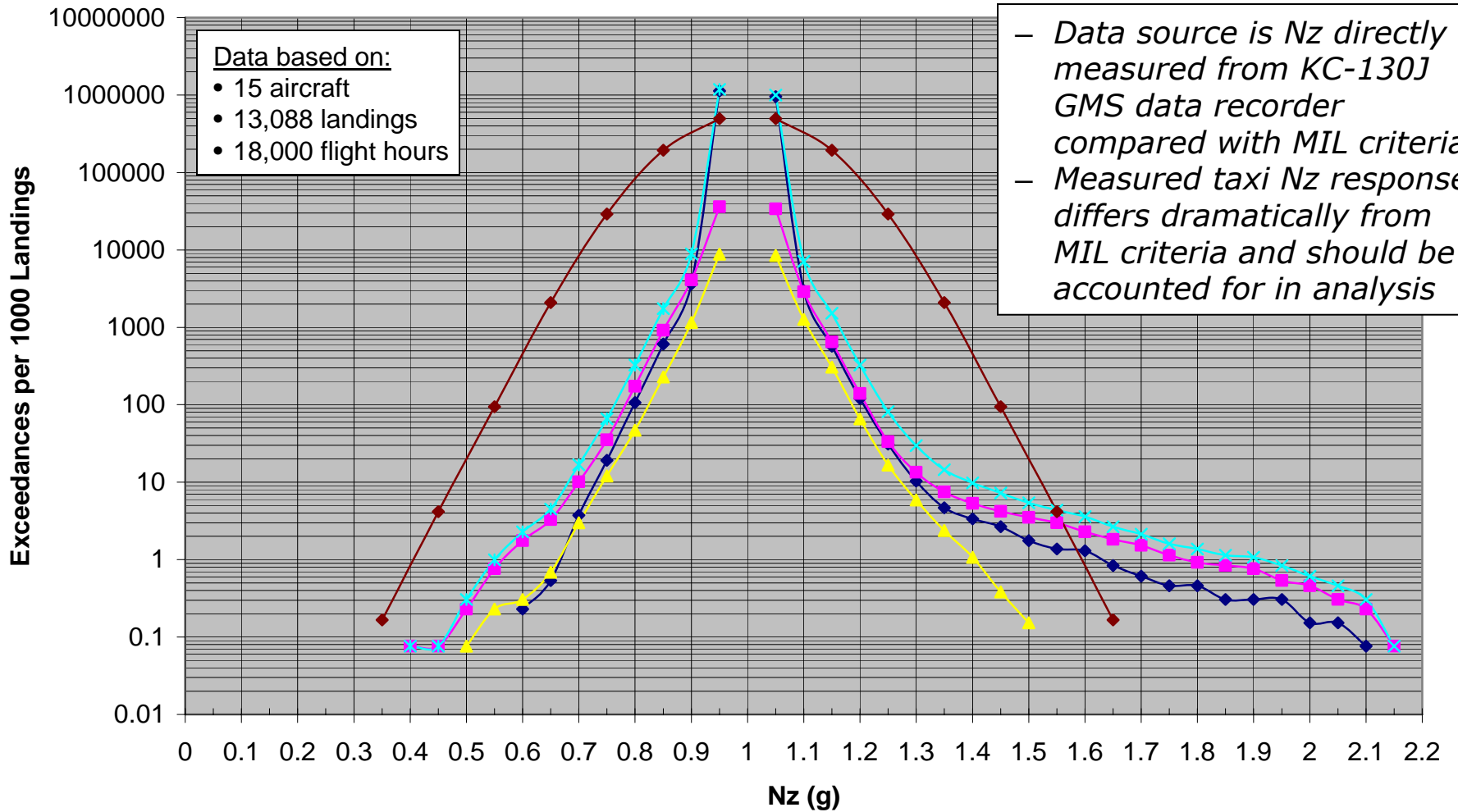


C-130 Gust Nz Exceedances Comparison



KC-130J Taxi Nz Exceedances

◆ Taxi Out
 ■ Taxi Rollout
 ▲ Take-off
 ✦ Taxi Total
 ◆ MIL-A-8866C(AS)

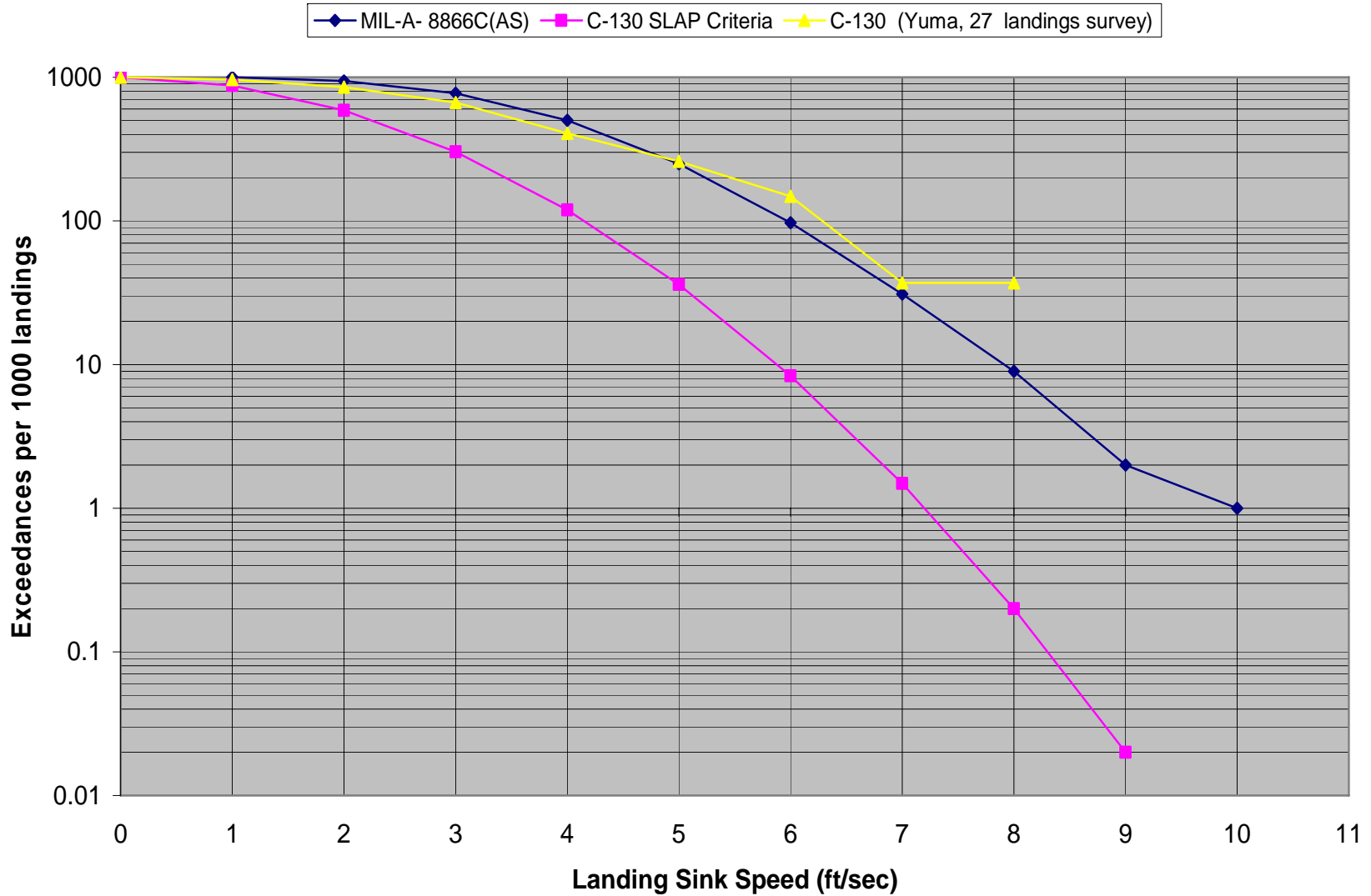


Landing Impact Criteria

- NAVAIR has performed video landing surveys at Cherry Point, NC; Yuma, AZ; and Gila Bend, AZ
- The purpose is as follows:
 - Provide a non-interference method to:
 - Collect operational data for the calculation of landing loads
 - Identify trends in aircraft landing performance
 - Vertical velocity (sink speed), horizontal velocity, and attitude are determined at touchdown



C-130 Landing Sink Survey (Yuma, 27 landings)



Conclusions

- Gust cycle clustering is a real phenomena and should be considered when assembling a stress spectrum
 - The criticality of crack growth due to gust loads dictates the development of realistic gust cycle placement methods
- Measured KC-130F/R/T and C-130T flight data have been analyzed to develop maneuver, gust, VH, VGH, weight distributions, missions profiles, and mission maneuver sequence criteria
- Measured KC-130J data have been analyzed to develop criteria for taxi, sink speed, lateral maneuvers, and ground handling
- The severity of maneuver loading and landing sink has been increased relative to baseline criteria
- High-fidelity flight data recorders and individual aircraft tracking are critical for realistic aircraft life management

Questions?