# **478th Aeronautical Systems Wing**

Delivering 21<sup>st</sup> Century Air Dominance!

### Aircraft Structural Integrity and the F-22

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Outline

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- Program Overview
- Why is ASIP Important?
- ASIP on the F-22
- Lessons Learned
- Way Forward
- Summary



(U.S Air Force photo by Staff Sgt Eric T. Sheler)



## **Program Overview**

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## **Program Overview** Aircraft Characteristics

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- Primary Function: Air dominance, multi-role fighter
- Wingspan: 44 feet, 6 inches
- Length: 62 feet, 1 inch
- Height: 16 feet, 8 inches
- Weight: 19,700 pounds
- Maximum Takeoff Weight: 83,500 pounds
- Power Plant: Two Pratt & Whitney F119-PW-100 turbofan engines
- Speed: 1,140 mph (Mach 1.72); supercruise at altitude
- Armament:
  - 1 M61A2 20-millimeter cannon
  - 2 AIM-9 infrared missiles in side weapon bay
  - 6 AIM-120 AMRAAMs in main weapon bay
    - or 2 GBU-32 JDAMs and 2 AIM-120 AMRAAMs



47/8 AES M



## **Program Overview** How goes it in 2007

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### **Program Overview** F-22 Program Schedule

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## **Program Overview Production Program of Record**

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100% of Production on Contract and Over 50% Delivered



### Program Overview F-22 Basing Plan

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- Delivering 21<sup>st</sup> Century Air Dominance
  - Our Vision: "To provide the warfighter with the most capable and available fighter in the world...the F-22"
- F-22 ASIP provides the means to achieve:
  - Flight safety and mission effectiveness
  - Proactive approach to force management
  - Risk mitigation

(www.af.mil, U.S. Air Force photo/Tech Sgt Justin D. Pyle)



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### Delivering 21<sup>st</sup> Century Air Dominance



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- ASIP is mandated by the Air Force
  - Required via AFI 63-1001 / AFPD 63-10 / Mil-Std 1530
  - The only integrity program covered by USAF policy
- Experience makes it very clear: "Do it"
  - Legacy USAF aircraft, early Raptor program

(www.af.mil, U.S. Air Force photo/Tech Sgt Justin D. Pyle)



**Proactive approach to force management** 

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- Ensures structural integrity throughout service life
  - Accurately accounts for fleet variation of usage and loads environment
- Enables flexible response to force management issues
- Reduces operational risk and maintenance impact
  - Supports planning of future maintenance actions
  - Minimizes maintenance effort required for inspections
  - Improves accuracy of analytical predictions through repetitive verification
- Provides effective data base to support modernization changes
- Provides useful data for external programs (e.g. F-35)
- Reduces <u>life cycle cost</u>

(www.af.mil, U.S. Air Force photo/Tech Sgt Justin D. Pyle)

#### ASIP on the F-22 **Engineering/Production Concurrency** Delivering 21<sup>st</sup> Century Air Dominance! **47/8 AESW** 1998 1996 1997 1999 2000 2002 2003 2004 2005 2006 2008 2009 1995 2001 2007 Fatigue Spectrum Static Design Fatigue Design Update Loads Loads Full Scale Static and Durability EMD **Test Programs** EMD A/C PRTV PRTV II Lot 2 Static Design Update Lot 1 Lot 2 Production Lot 3 Fatigue Design Update Lot 3 Lot 4 Lot 4 Fatigue Test Update (Side of Body, Aft Boom, Lugs) **Concurrency of EMD and Production** Lot 5 led to first incorporation of full scale Lot 6 Lot 7 durability test fixes at Lot 4 (A/C 4062) Lot 8 Lot 9

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## ASIP on the F-22 Fatigue Test Experience





## ASIP on the F-22 F-22 Structures Management

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Plan early and Educate Stakeholders on Requirements





## ASIP on the F-22 Aircraft Usage

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•F-22's flown at high G's and heavy weight

•NzW = Vertical Load Factor x Aircraft Gross Weight: primary loads

driver for many structural critical locations

•Risk: Reduced aircraft availability; more inspections, more mods



NzW Exceedances

NzW at CG

**Fleet Usage Data Allows Proactive Management Actions** 



## **ASIP Progress on the F-22**

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- Task I Design Information
  - Complete

#### Task II – Design Analyses and Development Tests

- Open analyses: Final EMD spectrum, weight growth study, durability and damage tolerance updates, lug bore
- Open testing: Lug bore element tests, full scale frame tests, ground vibration test
- Open development: Non-destructive Inspection (NDI) technology
- Task III Full Scale Testing
  - Complete
- Task IV Force Management Data Package
  - Complete, will require periodic updates
- Task V Force Management
  - Ongoing until A/C retirement



(U.S. Air Force photo/Tech. Sgt. Shane A. Cuomo)



**Development/Production Concurrency** 

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- Program concurrency implemented to deliver F-22 operational capability early
  - 60 aircraft delivered prior to completion of fatigue test
  - Delayed incorporation of "in-line" production improvements
  - Configuration changes required for early aircraft

Lesson: Programs need to plan for retrofit activities post EMD to cover concurrency shortfalls

(U.S. Air Force photo by Kevin Robertson)



### Lessons Learned Building Block Component Testing

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- Forward boom assembly component test was reduced to streamline program and reduce costs
  - Lug bore and lug radii cracks found during full scale fatigue testing late in EMD
  - Structural analysis was correlated to sub-scale test specimens and did not adequately predict stress levels of the full scale structure
  - Originally planned component test would have identified problems in early EMD
- Lesson: Building block test would have identified problem early and prevented higher maintenance/quality control costs

(U.S. Air Force photo by Kevin Robertson)

### **Lessons Learned** Building block approach

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**Scale-up of Structural Castings** 

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- Large titanium castings were not adequately developed prior to production transition
  - Manufacturing scale-up issues not found until production
    - Rigorous NDI implemented in production to assure quality
    - New damage tolerance analysis methodology developed
    - Wing side of body casting redesigned as a forging
- Lesson: Manufacturing scale-up of proven technologies not exempt from rigorous steps to successfully transition into production



#### **Materials/Analysis Methodology**

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- Material selection challenged analysis methodology and NDI capability
  - Methodology
    - New analytical approaches required for Beta titanium due to large grain size
    - Poor life predictions using existing crack modeling techniques
  - NDI
    - Response variation due to grain size and surface treatments
    - Geometries and access restrictions (systems and Low Observables Coatings) drove new technology development
- Lesson: New materials need to be fully characterized and methodology adapted to handle new materials and geometries. NDI capabilities need to be a consideration during design.

(U.S. Air Force photo by Kevin Robertson)

#### **Design for Inspectability**



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- Panel corrosion discovered on fielded aircraft after extended exposure to moist environment
  - Galvanic re-action with gap filler and exterior aluminum panels
    - Corrosion confined to specific aluminum panel skin joint areas
    - Structural integrity implications if not corrected
  - Multi-faceted repair & prevention plan underway

 Lesson: Full scale system exposure to relevant operational environments early in test program would have drove earlier design/retrofit mitigation procedures and reduced "clean-up" costs



## Lessons Learned Corrosion Prevention Plan

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GASKE

GAP FILLER

AI

BULKHEAD

#### Corrosion mechanism well understood & characterized

- Galvanic action with gap filler, exterior aluminum panels & panel attach areas
- Corrosion confined to specific aluminum panel skin joint areas
- Multi-faceted repair & prevention plan executing since CY06
  - Low program risk





AI

New Seal Frame

Ti

**47/8 AESW** 

GASKET

SEAL FRAME

## Lessons Learned Corrosion Prevention Plan - Near Term

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## Way Forward Structural Retrofit Programs

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- Structural Retrofit Programs (SRP) will bring early production aircraft up to full service life configurations
  - SRP 1
    - In progress at depots
    - Addresses the most time critical locations
  - SRP 2
    - In development and planning will address remaining critical locations
    - Cost Benefit Analysis approach to prioritize retrofit locations

(U.S. Air Force photo/Tech. Sgt. Ben Bloker)



## Way Forward SRP Development

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- Continue F-22 test programs to support SRP
  - Lug bore element testing completed
    - Successfully verified cold work expansion provided sufficient benefit to meet design service life requirement
  - Frame crack growth tests
    - Component frame tests to evaluate crack growth data on wing attach frames not obtained during full scale fatigue test
  - Frame peening evaluation
    - Additional component frame testing to verify and select best fatigue enhancement method to extend service life of wing attach frames

<sup>(</sup>U.S. Air Force photo/Tech. Sgt. Ben Bloker)



## Way Forward Specific ASIP II Tasks

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- F-22 ASIP Projects to complete
  - Continuation of coupon testing to further understand F-22 material properties
  - Development and verification of new analysis methodology
    - Crack growth through residual stress zones
    - Shallow gradient crack growth
  - Structural loads and analysis update
    - Supports modeling effectiveness
  - Weight Growth assessments
    - Loads, Flutter & dynamics
    - Supports new capability insertions
  - Organic NDI capability improvement

(U.S. Air Force photo/Tech. Sgt. Ben Bloker)



## **F-22 Modernization Plan**

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**47/8 AESW** 



#### **ASIP Supports Changes to Weapon System**



## Way Forward ASIP Task V Projects

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- ASIP TASK V Activities
  - High fidelity fleet usage tracking system
    - Capture impact to airframe due to modernization
    - Package inspections to maximize aircraft availability
    - Capture effects of base usage variation (Ex- training fleet)
  - Structural certification of upgrades and new stores capability
  - Assessing risk to structural baseline due to modernization changes

(U.S. Air Force photo/Tech. Sgt. Ben Bloker)





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- ASIP is integral to delivering and sustaining 5<sup>th</sup> generation fighter capability to the USAF
- Multiple challenges remain in ASIP tasks II (development) and task V (sustainment) to ensure robust support of F-22
- Several valuable lessons learned can be gleaned for application to other programs
- F-22 program has a well defined way forward to assure safety and structural integrity for the service life of the weapon system

# ASIP provides risk reduction and cost avoidance while preserving safety of flight

(U.S. Air Force photo/Tech. Sgt. Ben Bloker)