



Tim Fallon USAF ASIP Conference – Dec 5, 2007



Design Specification

Air Vehicle Service Life

Ninety percent of all delivered JSF Air Vehicles, by variant, shall achieve either 30 years of operation or 8000 flight hours

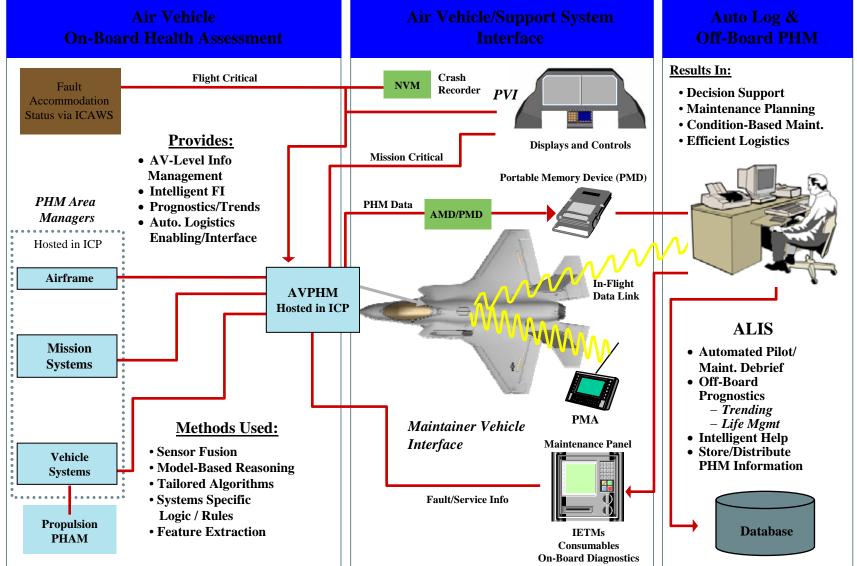


SPHM Goals

- Minimize maintenance while maintaining safety
- Eliminate scheduled inspections Goal is oncondition maintenance
- Achieve condition based maintenance at minimum cost
- Individual aircraft tracking
- Fully automated
- Support structural prognostics
- Minimize dedicated SPHM sensors
- SPHM integral part of AVPHM and OBPHM
- Capability to accept emerging technologies



PHM Architecture



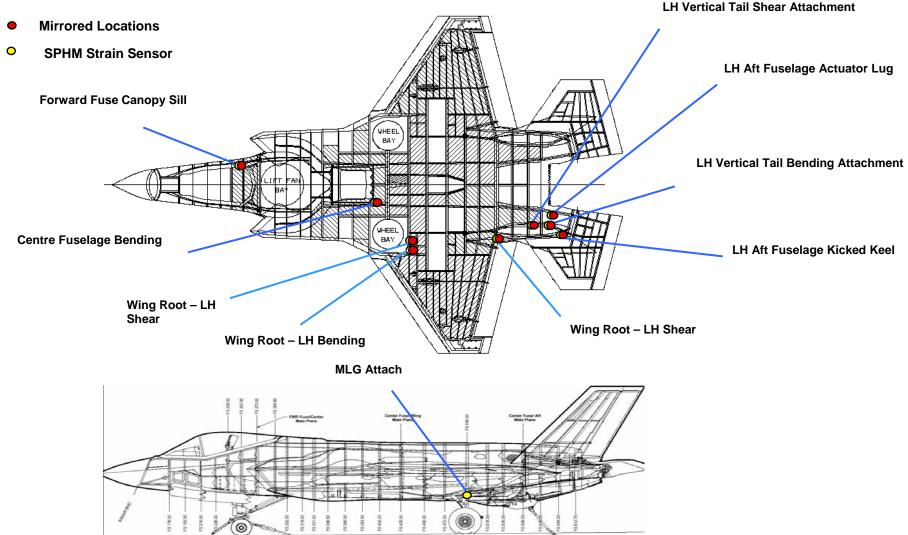


SPHM Functions

- Operational Loads Measurement
 - Strain gage
 - Parametric models
 - Safe-Life and Damage Tolerant models
- Structural Overload Measurement
- Auxiliary Structural Data collection
- Corrosion Environment Monitoring



SPHM Strain Sensor Locations (STOVL)





Strain Sensors

- JSF philosophy strain sensors are for model development, verification and refinement
 - All SDD and LRIP Aircraft with strain sensors
 - 10% of remaining aircraft with strain sensors
- One side of aircraft instrumented
 - Other side validated with loads aircraft instrumentation
- Can be in difficult to access locations
 - Primary and Backup gages
 - Will not be maintained operate until they expire
- All aircraft will be tracked primary means is parametric equations and dynamic models



Corrosion Sensors

- Replace scheduled inspections with "on condition" inspections
- Sensors in 2 locations on SDD aircraft
 - Demonstrate reliability and refine models
 - Growth capability exists as needs emerge
- Sentinel Resistance sensors
 - Resistance changes as pre-calibrated strips on sensor begin to corrode
 - Tied into aircraft bus to automatically record data
- Looking at multi-variate sensors for future



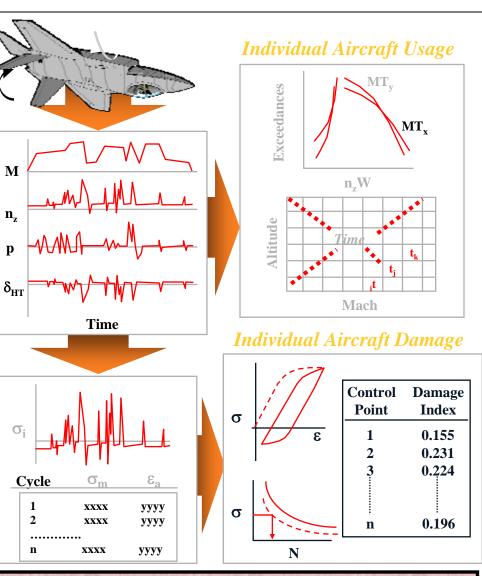
Corrosion Sensors





Individual Aircraft Tracking

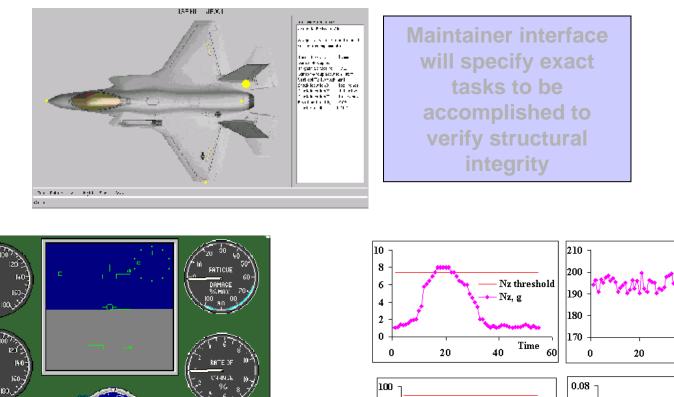
- SPHM Operational Loads Monitoring (OLM)
 - SPHM Area Manager Fed Various Flight Parameters
 - Most Data Sources SOF
 - Time History Captured
 - Parameter Cycle Counting and Usage Statistics Calculated by SPHM Area Manager
 - Fatigue Life Expended for Control Points Tracked
 - Results stored for ALIS download and further force management
 - Future updates are table driven, not OFP changes

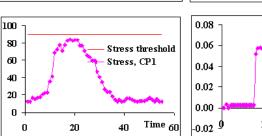


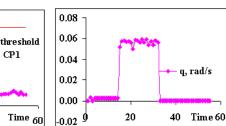
SPHM Architecture Allows 100% Data Capture vs. 40% or less for Legacy A/C



Structural Event Monitoring







– U, m/s

40

Time 60

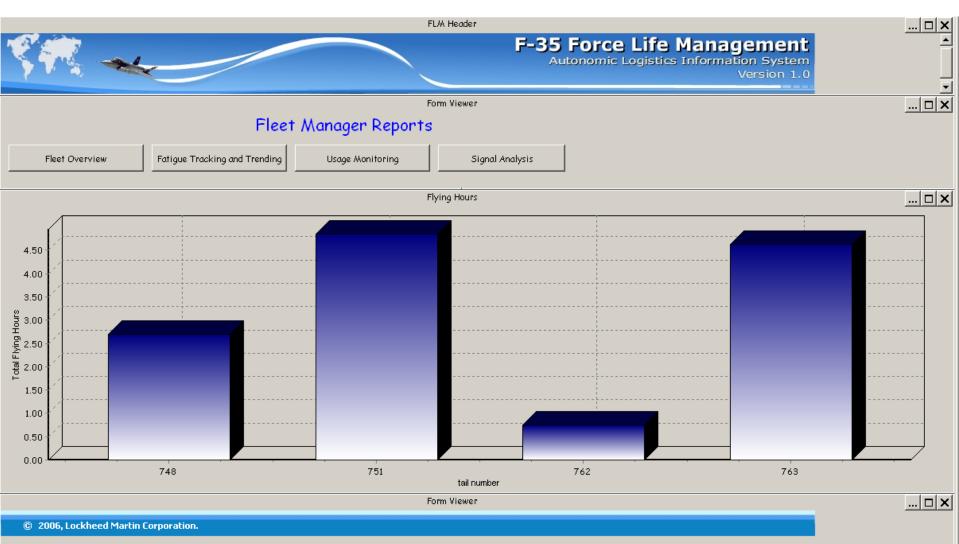
Parameter recording and display available for engineering analysis

Run Stop Time into Fight (S) 0.0 101530 F Lomporient

Flight recreation visualization aids understanding of damaging maneuvers

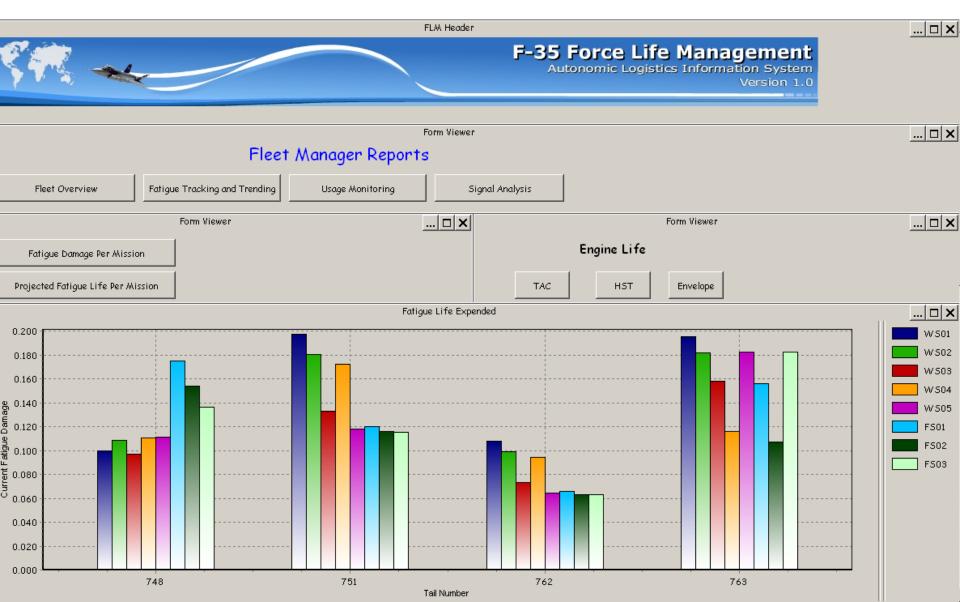


Sample FLM Screen



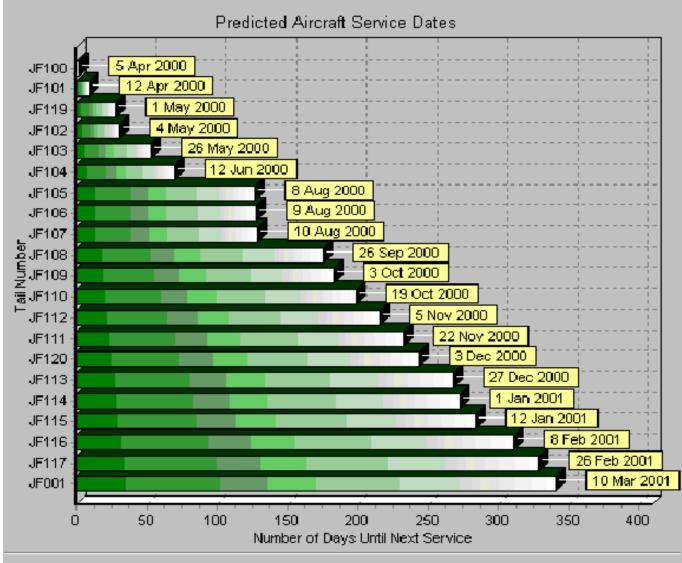


Sample Fleet Manager Screen





Force Life Management Capability



Close



Corrosion Model Concept

😻 ICAMS Analysis - Mozill	Firefox	хI
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Back Forward Reloa	Stop http://localhost:8080/icams/analyse?jsessionid=X1110933528623	
😤 Customize Links 😤 Free Hotmail 🔅 Windows Marketplace 😤 Windows Media 😤 Windows 😤 Internal Web		
The history is 2 months	(88 days) on ground and 32 days in the air.	
The following are the accumulated calculations for duration 3 months+ (120 days).		
	1mm 2mm	
Sloping Longeron	(Repair) (Replace) 638 μm	
Centre Wing	559 μm	
Outer Wing	424 μm	
Vertical Tail	473 μm	
Forward Avionics Bay	406 μm	
Under Deck - Rear	509 μm	
Under Deck - Forward	489 μm	
Upper Wing Surface	445 μm	
Fuselage	390 μm	
Main Landing Gear Ba	801 µm	
The history sequence w	κ.	
	DW - 4 hours' (A97-005-001)	
1 'On ground, wet base - 8 hours' (A97-005-002)		
1 'Fast Mission, Medium TOW - 4 hours' (A97-005-003) 1 'On ground dry base - 8 hours' (A97-005-004)		
1 'On ground dry base Done	- X hours' (A97-005-004)	
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Off Board Tools

- Assess Material Condition
- Anomaly and Failure Resolution System
- Knowledge Discovery
- Force Life Management



Future Technologies

- Crack detection and monitoring – CVM, MWM technologies
- Structural Integrity Prognosis System (SIPS)
- Airframe Reliability and Risk Assessment



Summary

- JSF embracing tenets of CBM+
- Data needs to support Assess Material Condition and Prognostics are high
 - 100+ paramaters continuously recorded at up to 320 Hz
 - Current projection is 200MB per hour that aircraft is powered
- Data transfer and storage needs stretch current capabilities

